



The Higher-Order
Unscented
Estimator

Zvonimir
Stojanovski

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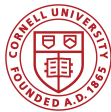
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A-Exam Presentation

Zvonimir Stojanovski

November 16, 2020



Motivation



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New imaging missions require increasingly accurate state estimation

- Driven by high-resolution optical instruments
- Need to know position, attitude, camera focus, etc.
- Required for controls, image processing, etc.

How can we improve state estimation for future missions?

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Cross-Calibration using Image Decomposition



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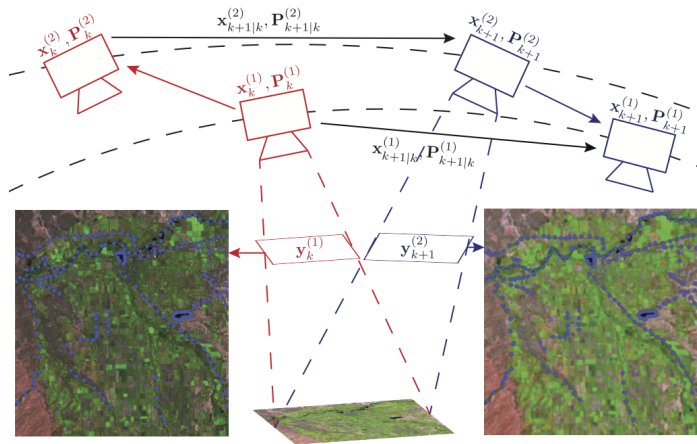
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D. Savransky (2020)



Spacecraft State Estimation



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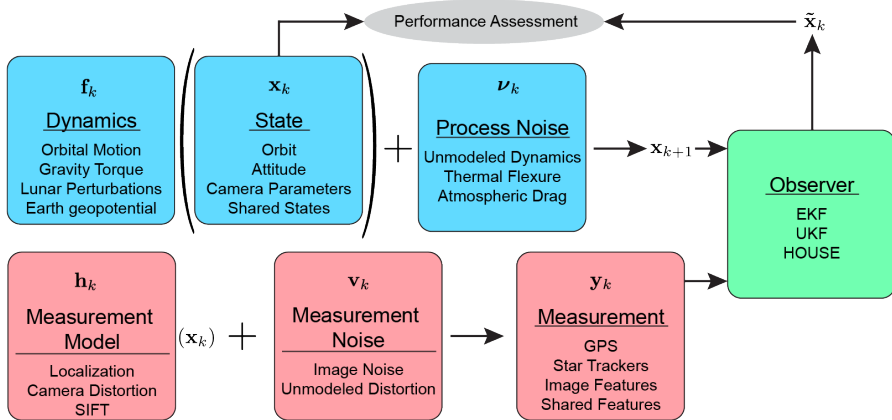
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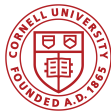
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D. Savransky (2020)



Challenges: Nonlinear Dynamics & Non-Gaussian Noise



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- Optimal filter does not exist in general case
- Approximations must be used
- Filter must be tractable in real time without sacrificing accuracy



Non-Gaussian Distributions: Image Feature Location



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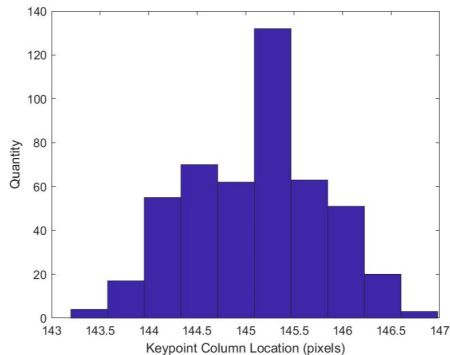
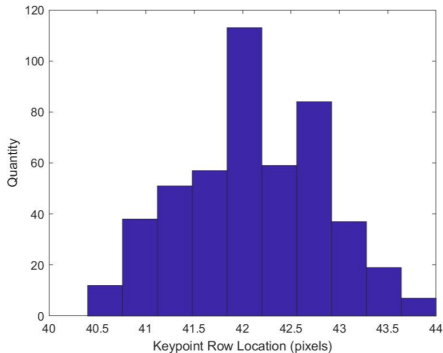
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J. Shapiro (2020)



Non-Gaussian Distributions: Image Feature Orientation & Scale



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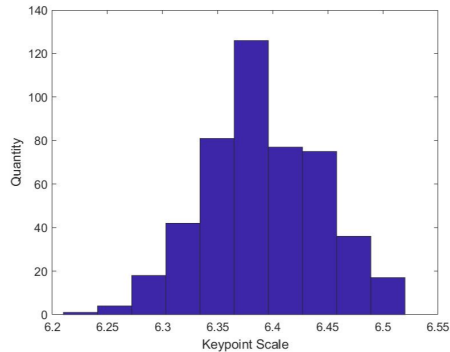
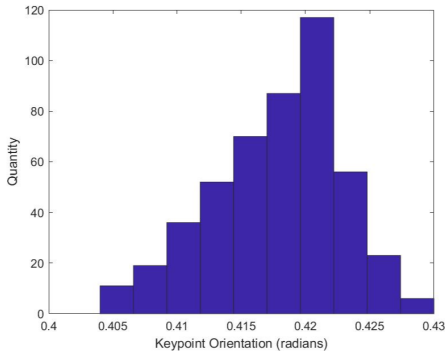
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J. Shapiro (2020)



The Nonlinear Filtering Problem



Estimate system state $\mathbf{x}(k)$ based on measurements $\mathbf{z}(k)$

Dynamics

$$\mathbf{x}(k+1) = \mathbf{f}(\mathbf{x}(k), \mathbf{u}(k), \mathbf{w}(k), k)$$

Measurement

$$\mathbf{z}(k) = \mathbf{h}(\mathbf{x}(k), \mathbf{u}(k), \mathbf{n}(k), k)$$

$\mathbf{w}(k)$ — Process noise

$\mathbf{f}(k)$ — State function

$\mathbf{n}(k)$ — Measurement noise

$\mathbf{h}(k)$ — Measurement function

$\mathbf{u}(k)$ — Control

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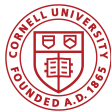
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Filter Operation



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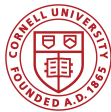
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Prediction

- Given distribution of $\mathbf{x}(k-1)$
- Find distribution of predicted state $\mathbf{x}(k|k-1)$

Update

- Given measurement $\mathbf{z}(k)$ and distribution of $\mathbf{x}(k|k-1)$
- Find distribution of updated state $\mathbf{x}(k)$



The Unscented Kalman Filter (UKF)



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- First proposed by Julier and Uhlmann (1997)
- Evaluates \mathbf{f} and \mathbf{h} at deterministic set of points
- Approximates distribution rather than functions
- Based on the *unscented transform*



The Unscented Transform

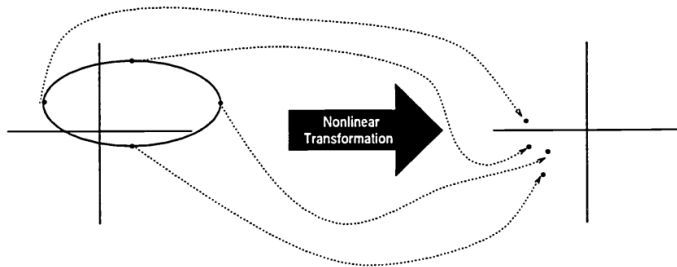


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- Given a random variable \mathbf{x} on \mathbb{R}^n with mean $\bar{\mathbf{x}}$ and covariance \mathbf{P}_{xx}
- How to approximate distribution of random variable $\mathbf{y} = \mathbf{g}(\mathbf{x})$?

One way is to take a set of points $\mathbf{x}^{(j)}$ and compute $\mathbf{y}^{(j)} = \mathbf{g}(\mathbf{x}^{(j)})$



Julier & Uhlmann (1997)

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The Unscented Transform



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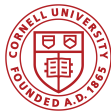
Sigma Points

$$\begin{aligned}\mathbf{x}^{(0)} &= \bar{\mathbf{x}} \\ \mathbf{x}^{(i)} &= \bar{\mathbf{x}} + \sqrt{n + \kappa} \mathbf{c}^{(i)} \\ \mathbf{x}^{(i+n)} &= \bar{\mathbf{x}} - \sqrt{n + \kappa} \mathbf{c}^{(i)}\end{aligned}$$

Weights

$$\begin{aligned}w_0 &= \frac{\kappa}{n + \kappa} \\ w_i &= \frac{1}{2(n + \kappa)} \\ w_{i+n} &= \frac{1}{2(n + \kappa)}\end{aligned}$$

- $\mathbf{c}^{(i)}$ are columns of $\sqrt{\mathbf{P}_{xx}}$
- Matrix square root usually taken to be Cholesky decomposition
- κ is a parameter for fine-tuning the filter



The Unscented Transform (UT)



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- Evaluate \mathbf{y} at sigma points for $j = 0, 1, \dots, 2n$:

$$\mathbf{y}^{(j)} = \mathbf{g}(\mathbf{x}^{(j)})$$

- Approximate mean and covariance of \mathbf{y} :

$$\bar{\mathbf{y}} = \sum_{j=0}^{2n} w_j \mathbf{y}^{(j)}$$

$$\mathbf{P}_{yy} = \sum_{j=0}^{2n} w_j (\mathbf{y}^{(j)} - \bar{\mathbf{y}})(\mathbf{y}^{(j)} - \bar{\mathbf{y}})^T$$

$$\mathbf{P}_{yx} = \sum_{j=0}^{2n} w_j (\mathbf{y}^{(j)} - \bar{\mathbf{y}})(\mathbf{x}^{(j)} - \bar{\mathbf{x}})^T$$

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Advantages of the UKF



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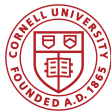
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- No Jacobians required, unlike extended Kalman filter (EKF)
- Fewer points required than particle filter (PF)



Limitations of the UKF



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- Determines only mean and covariance of state
- Sufficient for Gaussian distributions, but might not work with non-Gaussian effects

How can the UKF be generalized to represent a broader family of distributions?



Higher-Order Moments



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For a random variable x with mean μ and standard deviation σ :

- Skewness — measure of asymmetry

$$\gamma = \frac{\mathbb{E}[(x - \mu)^3]}{\sigma^3}$$

- Kurtosis — measure of “tailedness”

$$\kappa = \frac{\mathbb{E}[(x - \mu)^4]}{\sigma^4}$$

- Satisfy Pearson’s inequality

$$\kappa \geq \gamma^2 + 1$$

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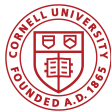
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Generalizing the UKF



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Ponomareva, Date, and Wang (2010) introduce two coefficients α and β to account for higher moments

Sigma Points

$$\begin{aligned} \mathbf{x}^{(0)} &= \bar{\mathbf{x}} \\ \mathbf{x}^{(i)} &= \bar{\mathbf{x}} + \alpha \sqrt{n} \mathbf{c}^{(i)} \\ \mathbf{x}^{(i+n)} &= \bar{\mathbf{x}} - \beta \sqrt{n} \mathbf{c}^{(i)} \end{aligned}$$

Weights

$$\begin{aligned} w_0 &= \frac{1}{2n} \\ w_i &= \frac{1}{\alpha(\alpha+\beta)n} \\ w_{i+n} &= \frac{1}{\beta(\alpha+\beta)n} \end{aligned}$$

- $\mathbf{c}^{(i)}$ are columns of $\sqrt{\mathbf{P}_{xx}}$
- α and β chosen to preserve marginal skewness and kurtosis *averaged* over components

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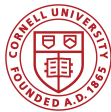
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Contributions



The Higher-Order Unscented Estimator

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A new extension of the unscented Kalman filter:
Higher-Order Unscented Estimator (HOUSE)

- Developed methodology and mathematical formalism
- Implemented and tested with simulations of dynamical systems

Publication: Z. Stojanovski and D. Savransky, “The Higher-Order Unscented Estimator,” submitted to the *Journal of Guidance, Control and Dynamics*.

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Other Relevant Work



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● Courses

- Intermediate Dynamics & Vibrations (MAE 5730)
- Feedback Control Systems (MAE 5780)
- Methods of Applied Mathematics I (MAE 6810)
- Advanced Dynamics (MAE 6700)
- Statistics (MATH 4720)
- Intelligent Sensor Planning & Control (MAE 6790)
- Advanced Astrodynamics (MAE 6720)
- Model-Based Estimation (MAE 6760) — *in progress*
- Teaching Assistant for System Dynamics (MAE 3260), Spring 2020



The Higher-Order Unscented Estimator (HOUSE)



The Higher-Order Unscented Estimator

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- A new extension of the UKF that captures third and fourth moments in n directions
- Builds on the method of Ponomareva et al.
- Instead of one value of α and β , we have n of each
- Sigma points are given by

$$\mathbf{x}^{(j)} = \begin{cases} \bar{\mathbf{x}} + \alpha_j \mathbf{c}^{(j)}, & 1 \leq j \leq n \\ \bar{\mathbf{x}} - \beta_j \mathbf{c}^{(j)} & n + 1 \leq j \leq 2n \\ \bar{\mathbf{x}} & j = 2n + 1 \end{cases}$$

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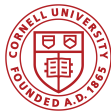
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HOUSE Prediction



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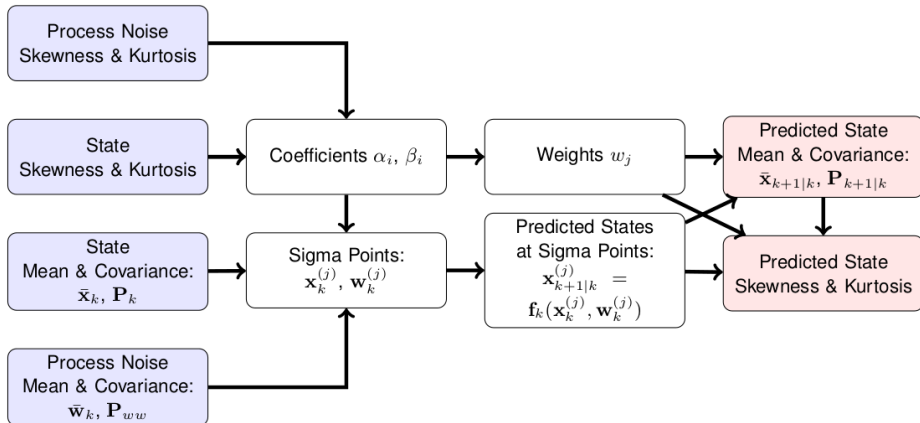
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For brevity, we denote \mathbf{P}_{xx} by \mathbf{P}



More



HOUSE Update



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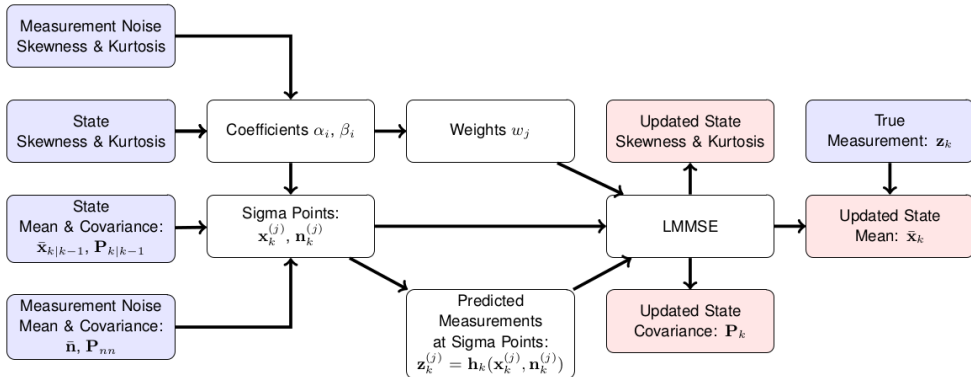
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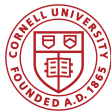
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Determining the Coefficients & Weights



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- Find coefficients α_j, β_j and weights w_j such that

$$\underbrace{\int_{\Omega} \phi(\mathbf{x}) p(\mathbf{x}) d\mathbf{x}}_{\text{Exact } E[\phi(\mathbf{x})]} = \underbrace{w_{2n+1} \phi(\bar{\mathbf{x}}) + \sum_{j=1}^n (w_j \phi(\bar{\mathbf{x}} + \alpha_j \mathbf{c}^{(j)}) + w_{n+j} \phi(\bar{\mathbf{x}} - \beta_j \mathbf{c}^{(j)}))}_{\text{Weighted sum approximation of } E[\phi(\mathbf{x})]}$$

for all functions ϕ in some family

- Effectively finding a cubature (multivariate quadrature) rule

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Standardization



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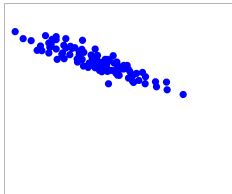
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For any random variable \mathbf{x} with mean $\bar{\mathbf{x}}$ and covariance \mathbf{P}_{xx} ,

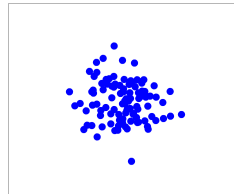
$$\tilde{\mathbf{x}} = (\sqrt{\mathbf{P}_{xx}})^{-1}(\mathbf{x} - \bar{\mathbf{x}})$$

has zero mean and identity covariance

\mathbf{x}



$\tilde{\mathbf{x}}$



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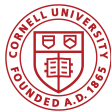
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Sigma Point Standardization



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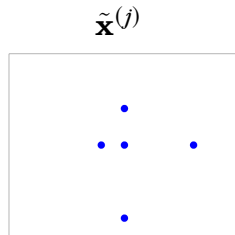
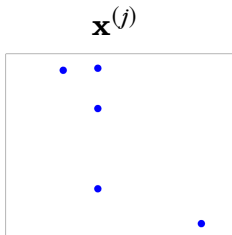
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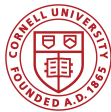
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$$\tilde{\mathbf{x}}^{(j)} = \begin{cases} \alpha_j \mathbf{e}^{(j)}, & 1 \leq j \leq n \\ -\beta_j \mathbf{e}^{(j)} & n+1 \leq j \leq 2n \\ \mathbf{0} & j = 2n+1 \end{cases}$$

- $\mathbf{e}^{(j)}$ are the standard basis vectors (orthonormal)
- α_j and β_j are the scaling coefficients (same as before)





Cubature Standardization



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Cubature rule with coefficients α_j, β_j and weights w_j

$$\int_{\Omega} \phi(\mathbf{x}) p(\mathbf{x}) d\mathbf{x} = w_{2n+1} \phi(\bar{\mathbf{x}}) + \sum_{j=1}^n (w_j \phi(\bar{\mathbf{x}} + \alpha_j \mathbf{c}^{(j)}) + w_{n+j} \phi(\bar{\mathbf{x}} - \beta_j \mathbf{c}^{(j)}))$$

holds for function ϕ if and only if standardized cubature rule

$$\int_{\tilde{\Omega}} \tilde{\phi}(\tilde{\mathbf{x}}) \tilde{p}(\tilde{\mathbf{x}}) d\tilde{\mathbf{x}} = w_{2n+1} \tilde{\phi}(\mathbf{0}) + \sum_{j=1}^n (w_j \tilde{\phi}(\alpha_j \mathbf{e}^{(j)}) + w_{n+j} \tilde{\phi}(-\beta_j \mathbf{e}^{(j)}))$$

holds for

$$\tilde{\phi}(\tilde{\mathbf{x}}) = \phi\left(\left(\sqrt{\mathbf{P}_{xx}}\right)\tilde{\mathbf{x}} + \bar{\mathbf{x}}\right)$$

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Determining the Coefficients & Weights



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Want the standardized cubature rule

$$\int_{\tilde{\Omega}} \tilde{\phi}(\tilde{\mathbf{x}}) \tilde{p}(\tilde{\mathbf{x}}) d\tilde{\mathbf{x}} = w_{2n+1} \tilde{\phi}(\mathbf{0}) + \sum_{j=1}^n (w_j \tilde{\phi}(\alpha_j \mathbf{e}^{(j)}) + w_{n+j} \tilde{\phi}(-\beta_j \mathbf{e}^{(j)}))$$

to hold exactly for

$$\tilde{\phi}(\tilde{\mathbf{x}}) = 1, \tilde{x}_i, \tilde{x}_i^2, \tilde{x}_i^3, \tilde{x}_i^4 \quad \text{for } i = 1, \dots, n$$



Determining the Coefficients & Weights



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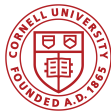
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System of equations for unknown coefficients α_i , β_i and weights w_i

$$\text{For } i = 1, \dots, n \quad \begin{cases} w_i \alpha_i - w_{n+i} \beta_i = 0 \\ w_i \alpha_i^2 + w_{n+i} \beta_i^2 = 1 \\ w_i \alpha_i^3 - w_{n+i} \beta_i^3 = \gamma_i \\ w_i \alpha_i^4 + w_{n+i} \beta_i^4 = \kappa_i \end{cases}$$

$$\sum_{j=1}^{2n+1} w_j = 1$$

where γ_i and κ_i are the skewness and kurtosis of the components of $\tilde{\mathbf{x}}$



Determining the Coefficients & Weights



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Solution

$$\alpha_i = \frac{\gamma_i + \sqrt{4\kappa_i - 3\gamma_i^2}}{2}$$

$$\beta_i = \frac{-\gamma_i + \sqrt{4\kappa_i - 3\gamma_i^2}}{2}$$

$$w_i = \frac{1}{\alpha_i^2 + \alpha_i\beta_i}$$

$$w_{n+i} = \frac{1}{\beta_i^2 + \alpha_i\beta_i}$$

$$w_{2n+1} = 1 - \sum_{j=1}^{2n} w_j$$

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Comparison of Sigma Points



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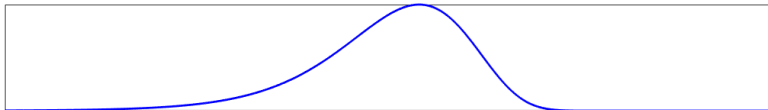
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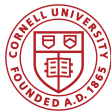


UKF Sigma Points



HOUSE Sigma Points





Properties of the Sigma Points



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- Due to Pearson's inequality, coefficients α_i and β_i must be real
- We have chosen the signs in the solution such that $\alpha_i, \beta_i > 0$
- All weights except w_{2n+1} guaranteed positive

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Mixed Terms



The Higher-Order Unscented Estimator

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- With our quadrature, all mixed monomials map to zero
- Exact for second-order terms (covariance)
- Effectively “drops” higher-order mixed terms in Taylor expansion of a function

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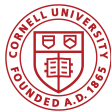
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The Weight at the Mean



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- w_{2n+1} is the only weight that might not be positive
- Not a valid probability distribution if negative — can be problematic
- Regardless of sign chosen for coefficients α_i and β_i ,

$$w_{2n+1} = 1 - \sum_{i=1}^n \frac{1}{\kappa_i - \gamma_i^2}$$

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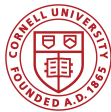
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Staying Positive



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- One solution is to scale γ_i and κ_i :

$$\gamma'_i = \gamma_i \sqrt{1 + \theta}$$

$$\kappa'_i = \kappa_i (1 + \theta)$$

where $\theta \geq 0$ is small

- Minimizes effects on lower-order moments
- Satisfies Pearson's inequality

$$\kappa'_i \geq \gamma_i'^2 + 1$$

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Selecting θ



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- To attain $w'_{2n+1} \geq \delta$, where $\delta \geq 0$, we can set

$$\theta = \begin{cases} 0, & S \leq 1 - \delta \\ \frac{S}{1-\delta} - 1 & S > 1 - \delta \end{cases}$$

where

$$S = \sum_{i=1}^n \frac{1}{K_i - \gamma_i^2}$$

- Minimal case $\delta = 0$ successfully used in simulations



HOUSE Constraints



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- For large values of kurtosis, sigma points may grow without bound
- To avoid this, impose the constraint, for some radius R

$$\|\mathbf{x}^{(i)} - \bar{\mathbf{x}}\| < R$$

- Other constraint — to ensure valid probability distribution

$$w_{2n+1} \geq 0$$

- Must hold for either original or tuned skewness and kurtosis

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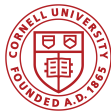
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HOUSE Constraints



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- Assume that \mathbf{x} has a radially symmetric distribution about $\bar{\mathbf{x}}$
- Then constraint inequalities reduce to

$$\sigma\sqrt{\kappa} < R$$

and

$$\frac{n}{\kappa} \leq 1$$

σ — Standard deviation

κ — Kurtosis

n — State dimension

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HOUSE Constraints



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- The range of κ where HOUSE can reliably operate is

$$n \leq \kappa < \frac{R^2}{\sigma^2}$$

- Expect the filter to perform better when this range is wider, i.e.,

$$n\sigma^2 \ll R^2$$

- Suggests that this filter is better for lower-dimensional systems

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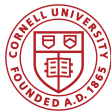
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The Pearson Type IV Distribution



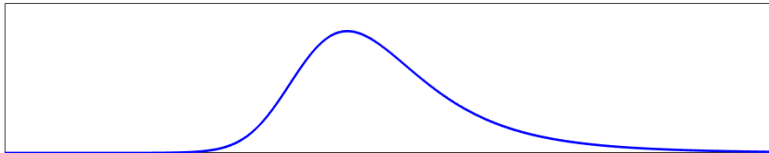
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Probability density function on $(-\infty, \infty)$

$$p(x) \propto \left(1 + \left(\frac{x - \lambda}{a}\right)^2\right)^{-m} \exp\left(-\nu \arctan\left(\frac{x - \lambda}{a}\right)\right)$$

- Free parameters $a > 0$, $m > 1/2$, λ , and ν
- Uniquely determined by first four moments



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Applications of Pearson Type IV



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- Wind shear fluctuations (Ramsdell, 1978)
- Fluctuating pressure on aircraft skin panels (Steinwolf & Rizzi, 2012)
- Solar wind intensity (Krafft, Gauthier & Volotkin, 2019)



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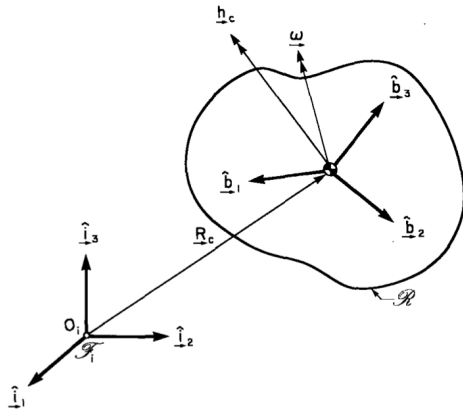
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“Rigid body \mathcal{R} with body-fixed frame \mathcal{F}_b and inertial frame \mathcal{F}_i .” (Hughes, 2004)



Rigid Body Example: Dynamics & Measurements



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- Equations of motion

$$I_1 \dot{\omega}_1 = (I_2 - I_3) \omega_2 \omega_3 + \tau_1$$

$$I_2 \dot{\omega}_2 = (I_3 - I_1) \omega_3 \omega_1 + \tau_2$$

$$I_3 \dot{\omega}_3 = (I_1 - I_2) \omega_1 \omega_2 + \tau_3$$

- τ_1, τ_2, τ_3 are random external disturbance torques (process noise)
- $I_1 = 400 \text{ kg} \cdot \text{m}^2$, $I_2 = 200 \text{ kg} \cdot \text{m}^2$, and $I_3 = 100 \text{ kg} \cdot \text{m}^2$
- Only ω_1 is measured directly, with 10 Hz sampling rate



Rigid Body Example: Initial Conditions



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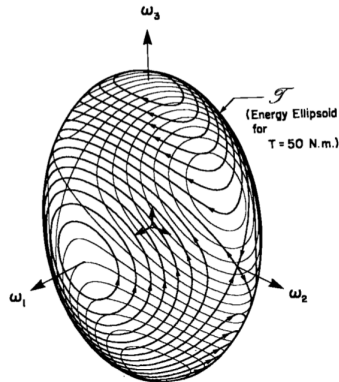
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- Test two sets of initial conditions
 - Major axis spinner:
 $\omega_1 = 0.01 \text{ rad/s}, \omega_2 = \omega_3 = 0$
 - Intermediate axis spinner:
 $\omega_2 = 0.01 \text{ rad/s}, \omega_1 = \omega_3 = 0$
- Both are stationary points
- First is stable in Lyapunov sense; second is unstable



Hughes (2004)



Rigid Body Example: Process Noise



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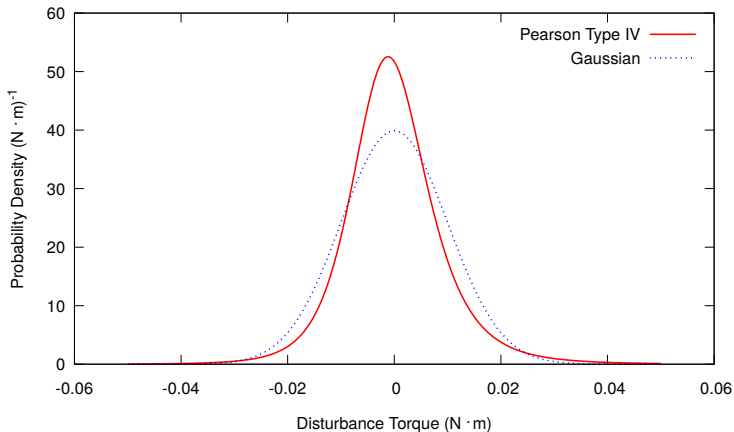
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Rigid Body Example: Measurement Noise



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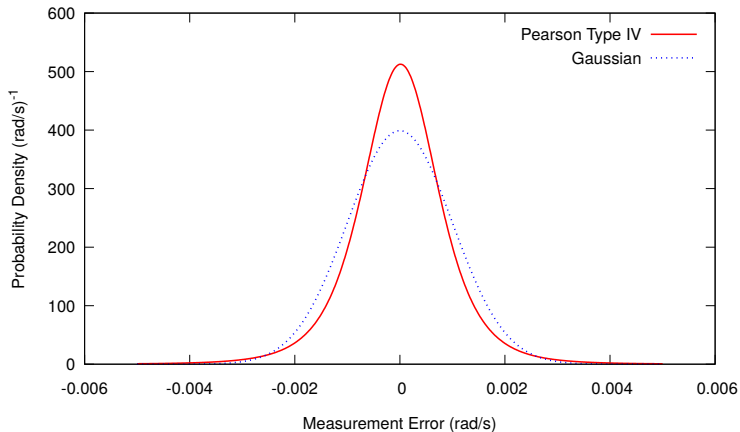
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Rigid Body Example: Estimation Error

Stable Spinner; Gaussian Noise



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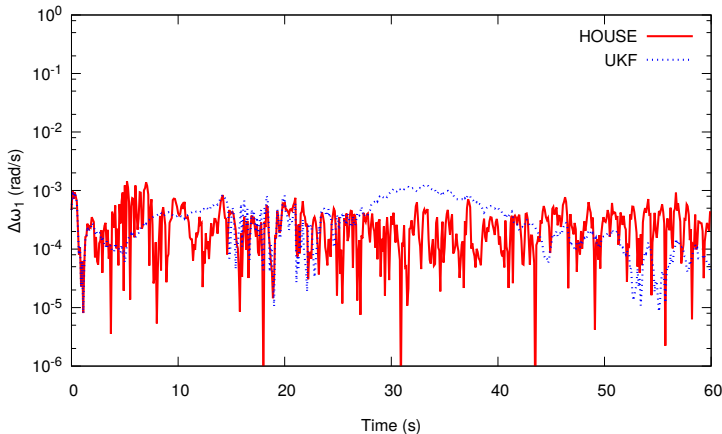
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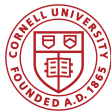
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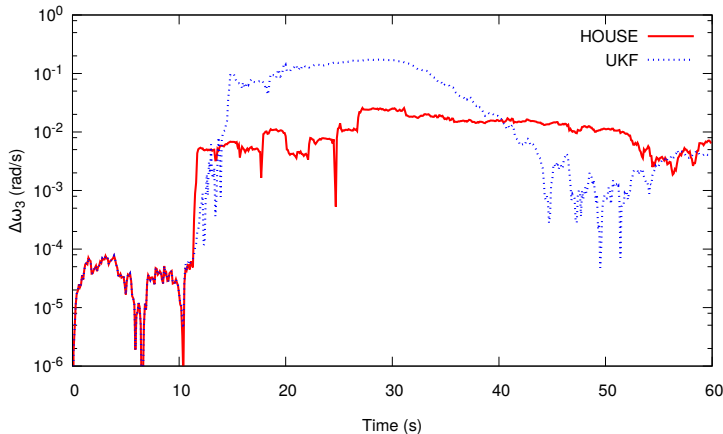
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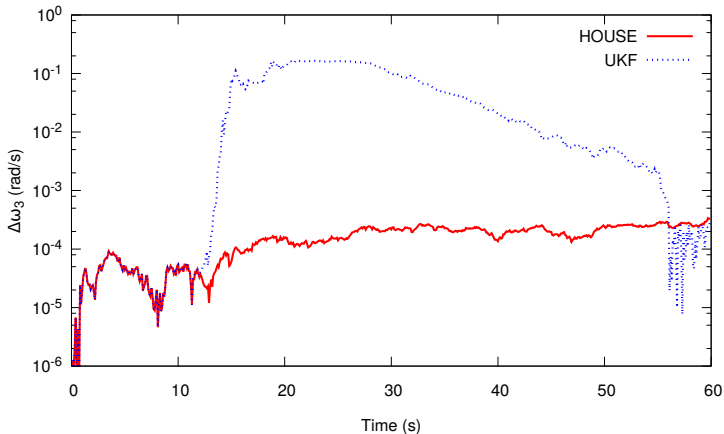
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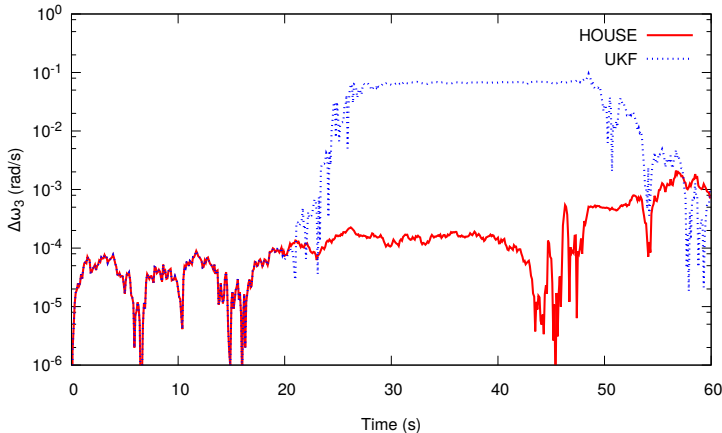
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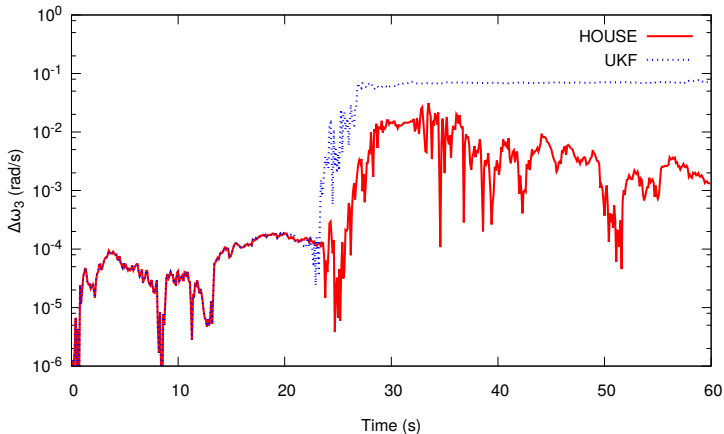
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Projectile Example: True Trajectories



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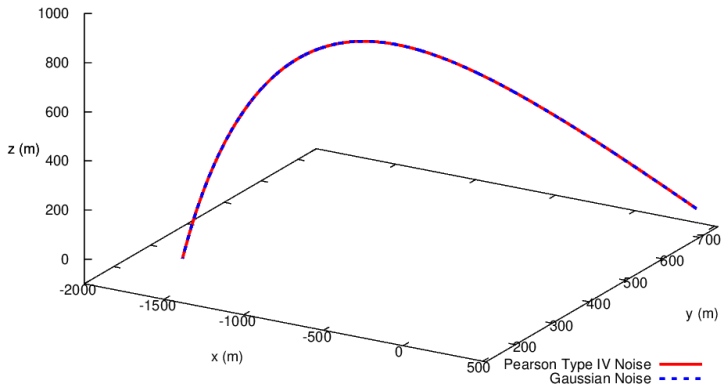
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Projectile Example: Dynamics



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- State consists of position (x, y, z) and velocity $(\dot{x}, \dot{y}, \dot{z})$
- Ballistic motion with drag

$$\ddot{x} = -bv\dot{x} + f_x$$

$$\ddot{y} = -bv\dot{y} + f_y$$

$$\ddot{z} = -bv\dot{z} + f_z - g$$

where

$$v = \sqrt{\dot{x}^2 + \dot{y}^2 + \dot{z}^2} \quad b = \frac{AC_D\rho}{2m} = 0.001 \text{ m}^{-1}$$

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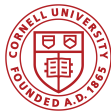
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Projectile Example: Measurements



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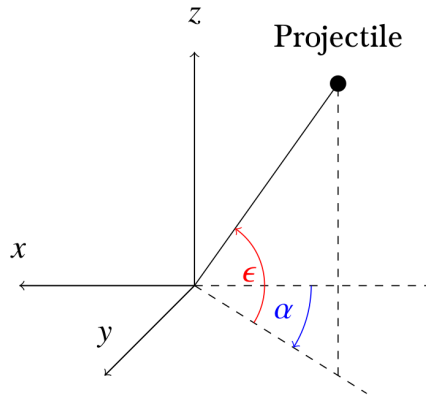
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- Line-of-sight only: Azimuth & Elevation

$$\alpha = \text{atan2}(y, -x) + n_\alpha$$

$$\epsilon = \text{atan2}\left(z, \sqrt{x^2 + y^2}\right) + n_\epsilon$$

- Measurements taken at 100 Hz
- Noise n_α , n_ϵ has standard deviation of 1 arc-minute





Projectile Example: Process Noise



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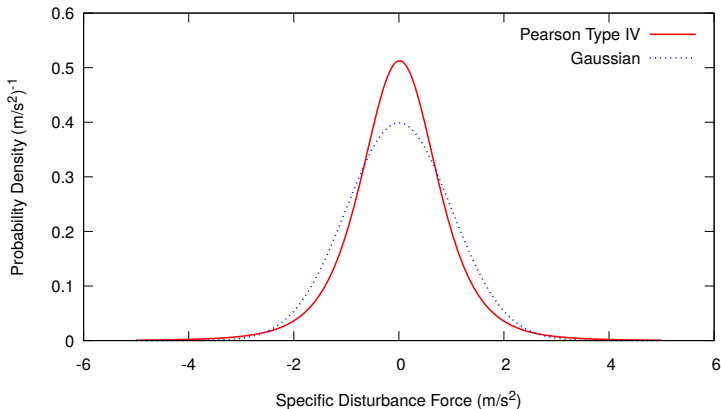
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Projectile Example: Measurement Noise



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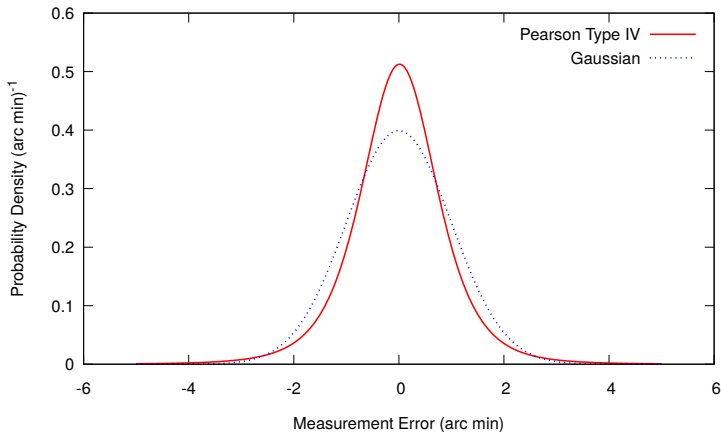
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Projectile Example: Estimation Error

Gaussian Noise



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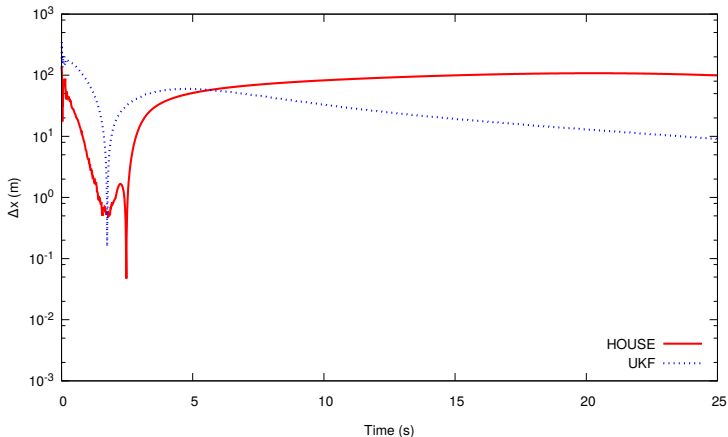
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Projectile Example: Estimation Error

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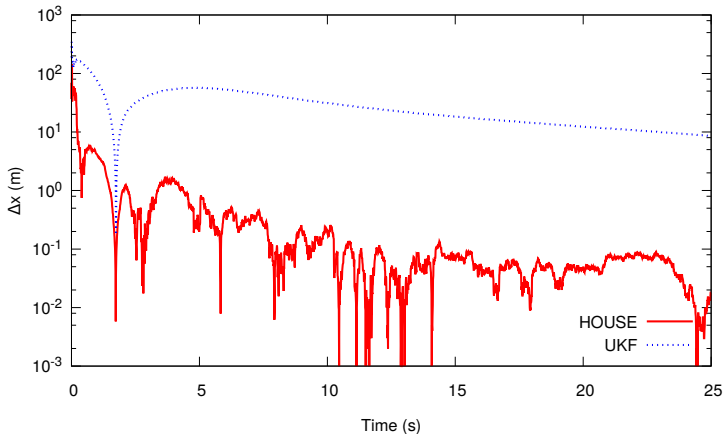
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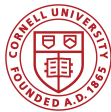
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Conclusions on HOUSE Performance



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- Provides significantly more accurate estimates than conventional UKF for heavy-tailed noise distributions
- Also appears to be more accurate for unstable systems
- Not as accurate for higher-dimensional systems with Gaussian noise



Future Work



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- Refining HOUSE—goals:
 - Make accuracy independent of system dimension
 - Higher accuracy for systems with Gaussian noise
- Analyzing computational complexity vs. accuracy trade-off
- Testing
 - High-fidelity spacecraft model
 - Key points extracted from images



Future Work



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- Next paper
 - HOUSE refinements
 - Deeper analysis and testing
- Exploring other filtering methods
 - Continuous-time prediction
 - Approximate Bayesian correction
- Spacecraft jitter estimation (NSTGRO proposal)

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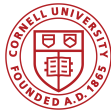
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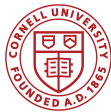
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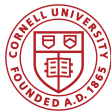
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The author acknowledges support for this work from the NASA Space Technology Research Grants Early Career Faculty program under NASA grant 80NSSC20K0068.



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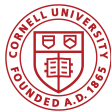
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HOUSE Operation: Prediction



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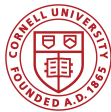
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- 1 For the augmented state $\mathbf{y}_P(k) = [\mathbf{x}(k)^T \quad \mathbf{w}(k)^T]^T$, generate the modified sigma points $\mathbf{y}_P^{(j)}(k) = [\mathbf{x}^{(j)}(k)^T \quad \mathbf{w}^{(j)}(k)^T]^T$ and weights $w_j(k)$
- 2 Propagate the state for each sigma point:

$$\mathbf{x}^{(j)}(k+1|k) = \mathbf{f}(\mathbf{x}^{(j)}(k), \mathbf{w}^{(j)}(k), k)$$

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- 3 Compute the predicted mean and covariance:

$$\bar{\mathbf{x}}(k+1|k) = \sum_{j=1}^N w_j \mathbf{x}^{(j)}(k+1|k)$$

$$\boldsymbol{\eta}^{(j)}(k+1|k) = \mathbf{x}^{(j)}(k+1|k) - \bar{\mathbf{x}}(k+1|k)$$

$$\mathbf{P}_{xx}(k+1|k) = \sum_{j=1}^N w_j \boldsymbol{\eta}^{(j)}(k+1|k) \boldsymbol{\eta}^{(j)T}(k+1|k)$$

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HOUSE Operation: Prediction



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- 1 Compute the standardized states at the sigma points:

$$\tilde{\mathbf{x}}^{(j)}(k+1|k) = \left(\sqrt{\mathbf{P}_{xx}(k+1|k)} \right)^{-1} \boldsymbol{\eta}^{(j)}(k+1|k)$$

- 2 Compute the skewness and kurtosis of the standardized state:

$$\gamma_i(k+1|k) = \sum_{i=1}^N w_j \tilde{x}_i^{(j)}(k+1|k)^3$$

$$\kappa_i(k+1|k) = \sum_{i=1}^N w_j \tilde{x}_i^{(j)}(k+1|k)^4$$

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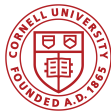
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Here, $\mathbf{z}(k)$ denotes the true measurement.

- 1 For the augmented state $\mathbf{y}_C(k) = [\mathbf{x}(k)^T \quad \mathbf{n}(k)^T]^T$, generate the modified sigma points $\mathbf{y}_C^{(j)}(k) = [\mathbf{x}^{(j)}(k)^T \quad \mathbf{n}^{(j)}(k)^T]^T$ and weights $w_j(k)$
- 2 Compute the measurement for each sigma point:

$$\mathbf{z}^{(j)}(k) = \mathbf{h}(\mathbf{x}^{(j)}(k), \mathbf{w}^{(j)}(k), k)$$

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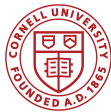
- 3 Compute the measurement mean and covariance:

$$\bar{\mathbf{z}}(k) = \sum_{j=1}^N w_j \mathbf{z}^{(j)}(k)$$

$$\mathbf{P}_{zz}(k) = \sum_{j=1}^N w_j (\mathbf{z}^{(j)}(k) - \bar{\mathbf{z}}(k)) (\mathbf{z}^{(j)}(k) - \bar{\mathbf{z}}(k))^T$$

$$\mathbf{P}_{xz}(k) = \sum_{j=1}^N w_j (\mathbf{x}^{(j)}(k|k-1) - \bar{\mathbf{x}}(k|k-1)) (\mathbf{z}^{(j)}(k) - \bar{\mathbf{z}}(k))^T$$

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- 1 Compute the LMMSE error and standardized state at the sigma points:

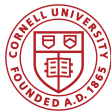
$$\boldsymbol{\epsilon}^{(j)}(k) = \mathbf{x}^{(j)}(k) - \bar{\mathbf{x}}(k|k-1) - \mathbf{P}_{xz}(k)\mathbf{P}_{zz}(k)^{-1}(\mathbf{z}^{(j)}(k) - \bar{\mathbf{z}}(k))$$

$$\tilde{\mathbf{x}}^{(j)} = \left(\sqrt{\mathbf{P}_{xx}(k)}\right)^{-1} \boldsymbol{\epsilon}^{(j)}(k)$$

- 5 Compute the skewness and kurtosis of the standardized state:

$$\gamma_i(k) = \sum_{i=1}^N w_j \tilde{X}_i^{(j)}(k)^3 \quad \kappa_i(k) = \sum_{i=1}^N w_j \tilde{X}_i^{(j)}(k)^4$$

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The constraint

$$\|\mathbf{x}^{(i)} - \bar{\mathbf{x}}\| < R$$

is equivalent to

$$\max_{i \in \{1, \dots, n\}} \left(\|\mathbf{c}^{(i)}\| \max(\alpha_i, \beta_i) \right) < R$$

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- Since

$$\max(\alpha_i, \beta_i) = \frac{|\gamma_i| + \sqrt{4\kappa_i - 3\gamma_i^2}}{2}$$

the constraint reduces to

$$\max_{i \in \{1, \dots, n\}} \left(\frac{|\gamma_i| + \sqrt{4\kappa_i - 3\gamma_i^2}}{2} \|\mathbf{c}^{(i)}\| \right) < R$$

- Needs further simplification

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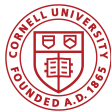
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- To have valid weights, the original or tuned γ_i and κ_i must satisfy

$$\sum_{i=1}^n \frac{1}{\kappa_i - \gamma_i^2} \leq 1$$

- This imposes another bound on the moments
- Again, needs simplification

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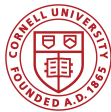
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- Assume that \mathbf{x} has a radially symmetric distribution about $\bar{\mathbf{x}}$
- Then, we have $\kappa_i = \kappa$, $\gamma_i = 0$, and $\|\mathbf{c}^{(i)}\| = \sigma$ for $i = 1, \dots, n$
- Constraint inequalities reduce to

$$\sigma\sqrt{\kappa} < R$$

and

$$\frac{n}{\kappa} \leq 1$$

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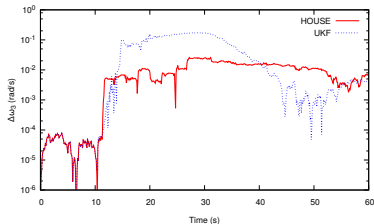
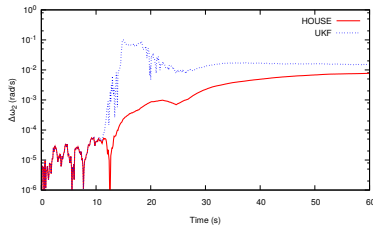
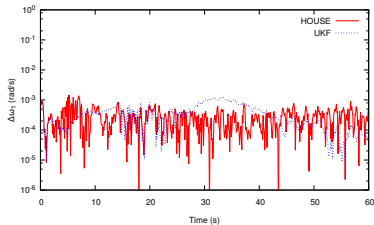
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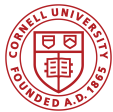
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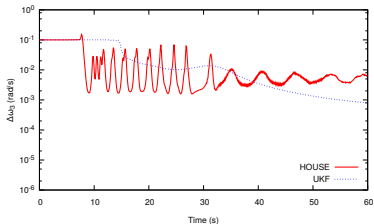
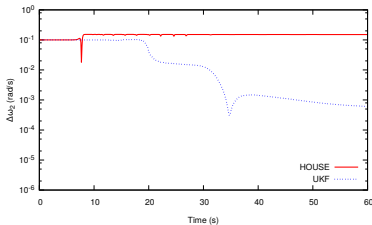
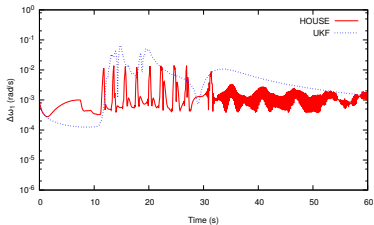
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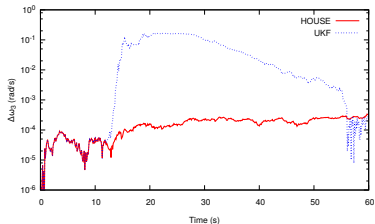
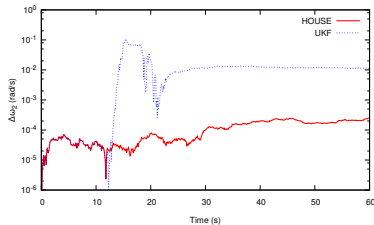
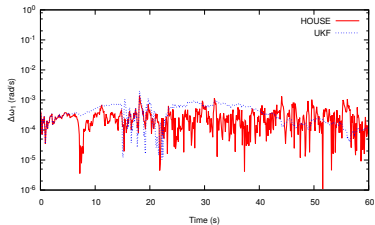
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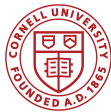
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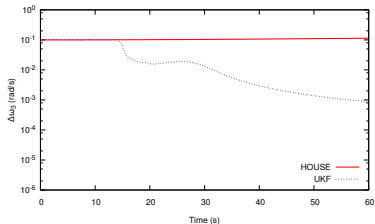
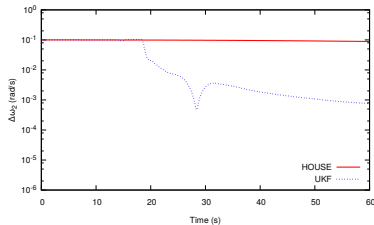
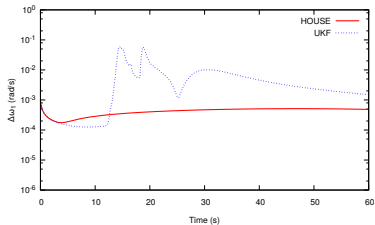
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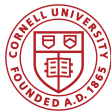
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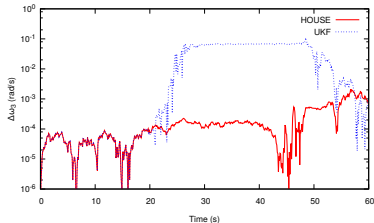
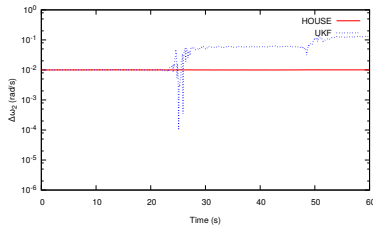
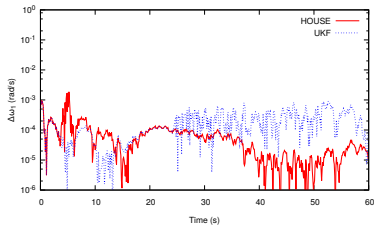
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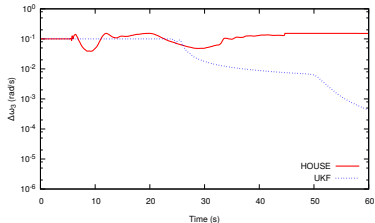
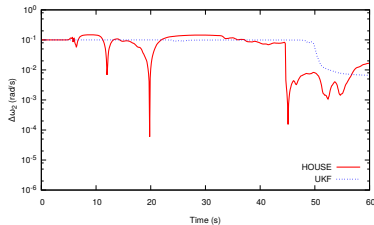
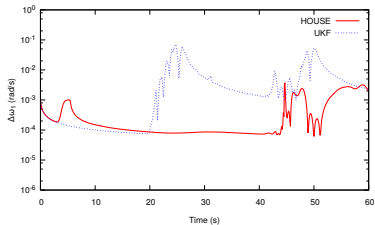
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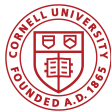
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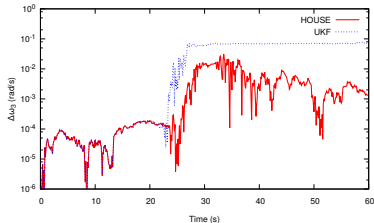
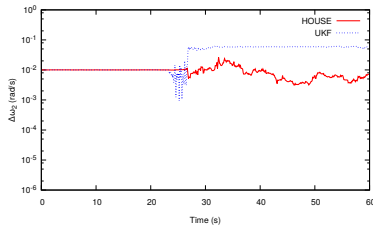
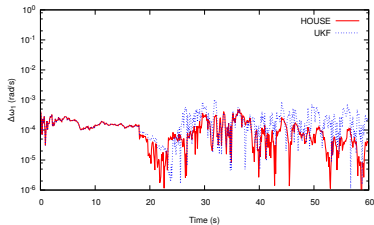
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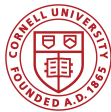
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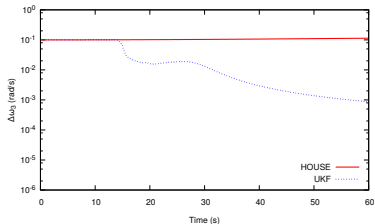
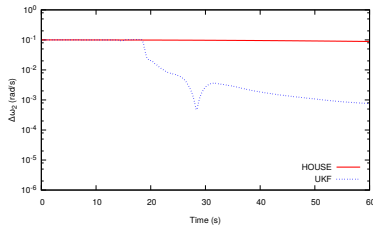
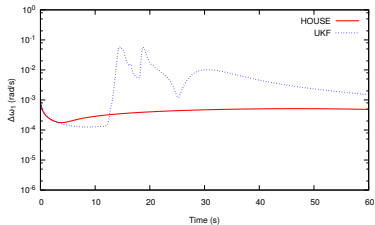
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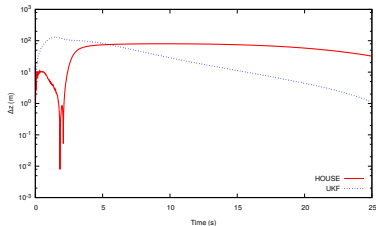
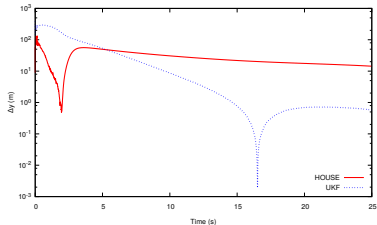
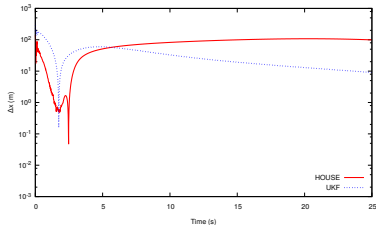
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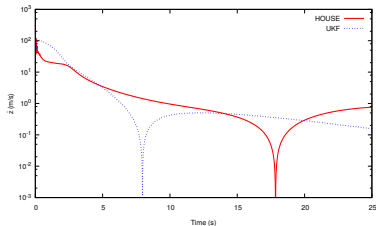
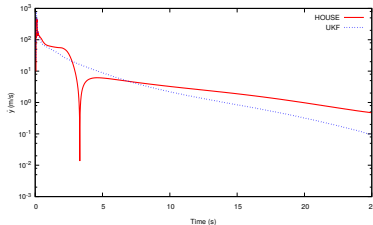
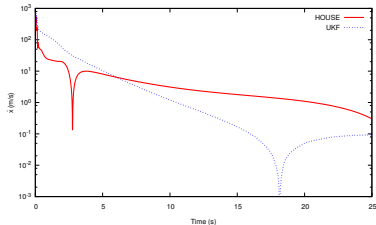
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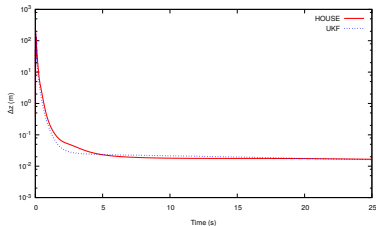
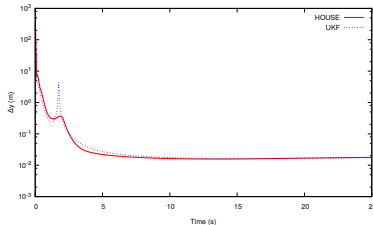
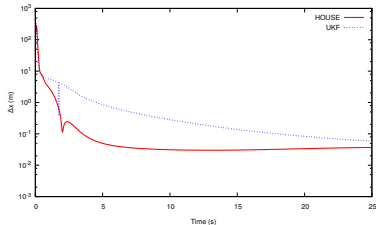
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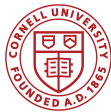
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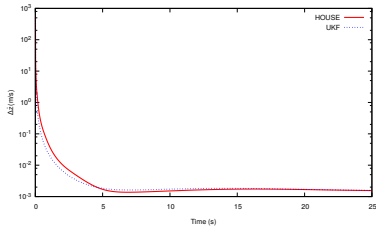
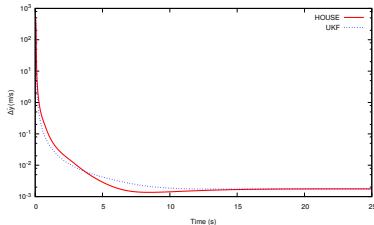
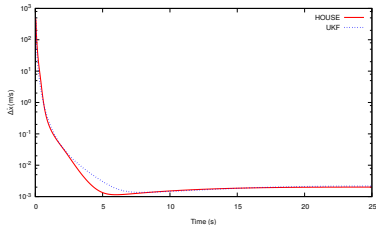
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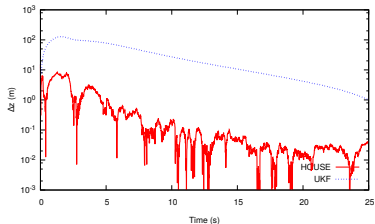
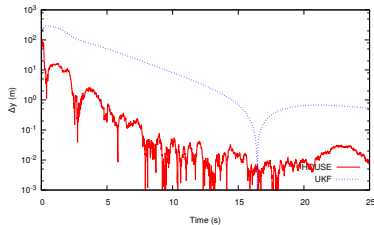
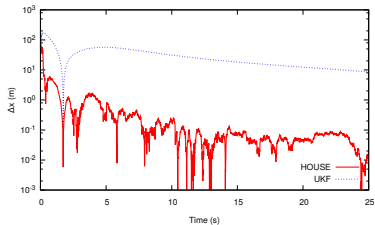
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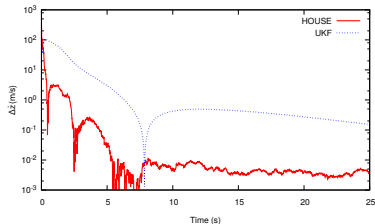
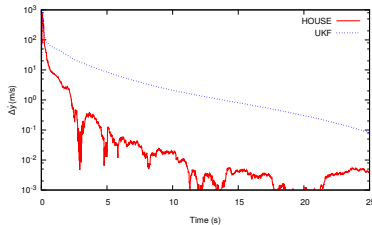
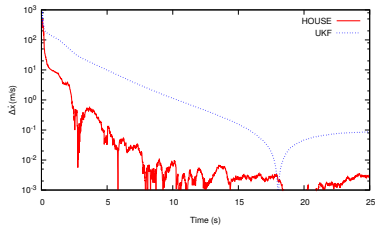
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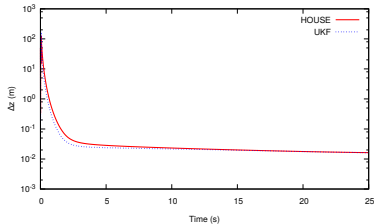
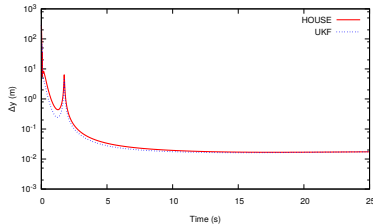
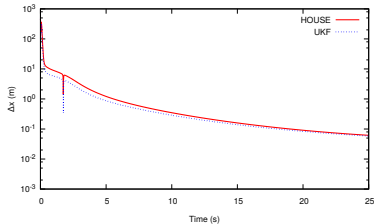
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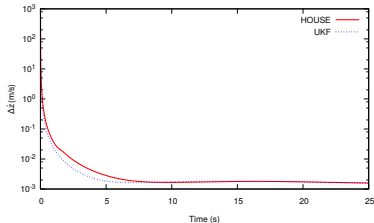
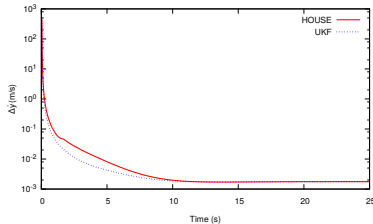
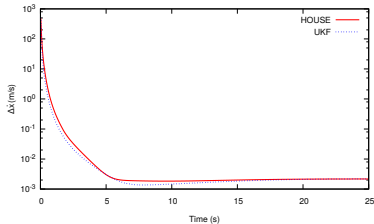
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Equivalence of Quadrature Rules



The Higher-Order Unscented Estimator

Zvonimir
Stojanovski

$$\begin{aligned}\int_{\Omega} \phi(\mathbf{x}) p(\mathbf{x}) d\mathbf{x} &= \int_{\tilde{\Omega}} \phi\left(\left(\sqrt{\mathbf{P}_{xx}}\right)\tilde{\mathbf{x}} + \bar{\mathbf{x}}\right) \tilde{p}(\tilde{\mathbf{x}}) d\tilde{\mathbf{x}} = \int_{\tilde{\Omega}} \tilde{\phi}(\tilde{\mathbf{x}}) \tilde{p}(\tilde{\mathbf{x}}) d\tilde{\mathbf{x}} \\ &= w_{2n+1} \tilde{\phi}(\mathbf{0}) + \sum_{j=1}^n (w_j \tilde{\phi}(\alpha_j \mathbf{e}^{(j)}) + w_{n+j} \tilde{\phi}(-\beta_j \mathbf{e}^{(j)})) \\ &= w_{2n+1} \phi(\bar{\mathbf{x}}) + \sum_{j=1}^n (w_j \phi(\bar{\mathbf{x}} + \alpha_j \mathbf{c}^{(j)}) + w_{n+j} \phi(\bar{\mathbf{x}} - \beta_j \mathbf{c}^{(j)}))\end{aligned}$$

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Equivalence of Quadrature Rules



The Higher-Order Unscented Estimator

Zvonimir Stojanovski

$$\begin{aligned}\int_{\tilde{\Omega}} \tilde{\phi}(\tilde{\mathbf{x}}) \tilde{p}(\tilde{\mathbf{x}}) d\tilde{\mathbf{x}} &= \int_{\tilde{\Omega}} \phi\left(\left(\sqrt{\mathbf{P}_{xx}}\right)\tilde{\mathbf{x}} + \bar{\mathbf{x}}\right) \tilde{p}(\tilde{\mathbf{x}}) d\tilde{\mathbf{x}} = \int_{\Omega} \phi(\mathbf{x}) p(\mathbf{x}) d\mathbf{x} \\ &= w_{2n+1} \phi(\bar{\mathbf{x}}) + \sum_{j=1}^n (w_j \phi(\bar{\mathbf{x}} + \alpha_j \mathbf{c}^{(j)}) + w_{n+j} \phi(\bar{\mathbf{x}} - \beta_j \mathbf{c}^{(j)})) \\ &= w_{2n+1} \tilde{\phi}(\mathbf{0}) + \sum_{j=1}^n (w_j \tilde{\phi}(\alpha_j \mathbf{e}^{(j)}) + w_{n+j} \tilde{\phi}(-\beta_j \mathbf{e}^{(j)}))\end{aligned}$$

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